



HEXAGON



Cradle

Cradle Progress Fixed Grid RANS TFG

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Software Cradle

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Topic 1 – Flap Deflection Studies

Numerical Method

Numerical method used in this simulation

- Overview

CFD software	scFLOW V2021
Method	Cell-centered finite volume method, Unstructured polyhedral, Density-based solver
Inviscid flux	Roe
Reconstruction	MUSCL, $\kappa=0.5$ for the meanflow equations
Limiter	van Leer-type Hishida limiter
Viscous flux	Alpha damping scheme
Gradient	Green-Gauss
Nonlinear solver	Implicit defect correction solver with the residual Jacobian derived exactly from a lower-order discretization with a local pseudo-time step
Linear solver	Block symmetric Gauss-Seidel
Turbulence model	SA-neg

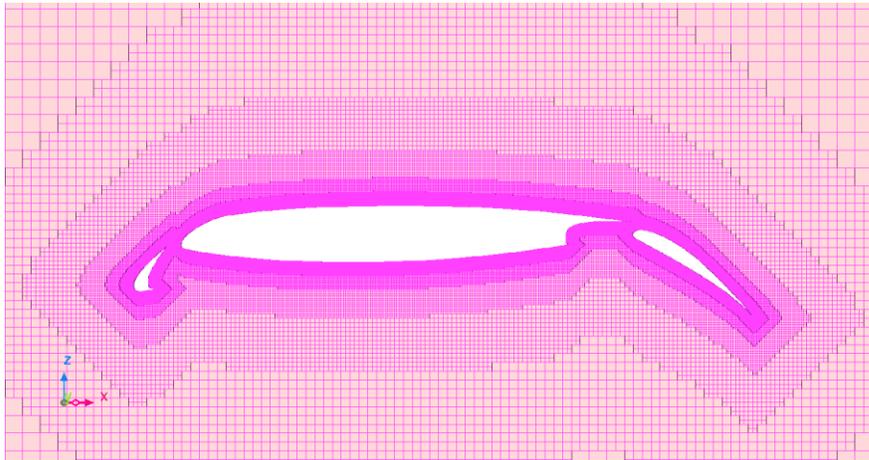
Reference for details

- Higo, Y., Nakashima, Y., Fujiyama, K., Irie, T., and Nishikawa, H., “RANS Solutions on Three Dimensional Benchmark Configurations with scFLOW, a Polyhedral Finite-Volume Solver,” AIAA Aviation 2020 Forum, AIAA Paper 2020-3029, 2020.

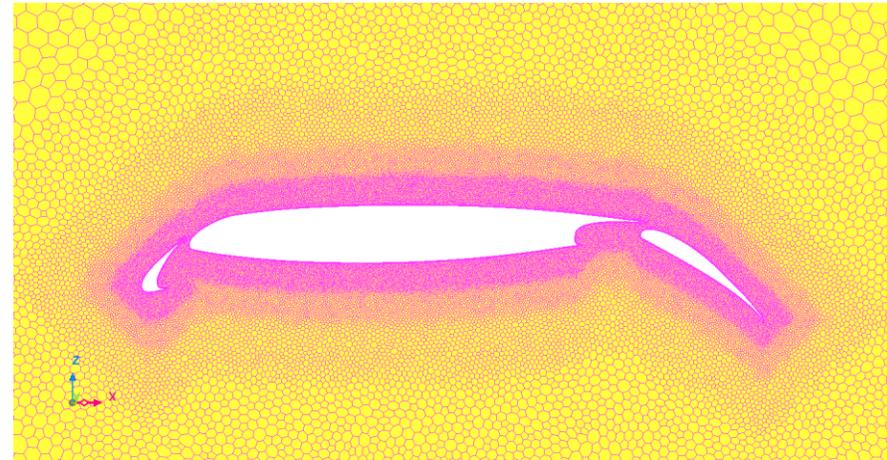
Numerical Method

Overview

- scFLOW is a comprehensive package including mesh generation.
- scFLOW generate Unstructured polyhedral mesh.
- In meshing, the element size is specified by "octant".
- Prism layers are automatically determined from the octant size of the wall surface. (Prism layers become thinner with finer octants)



octant



mesh

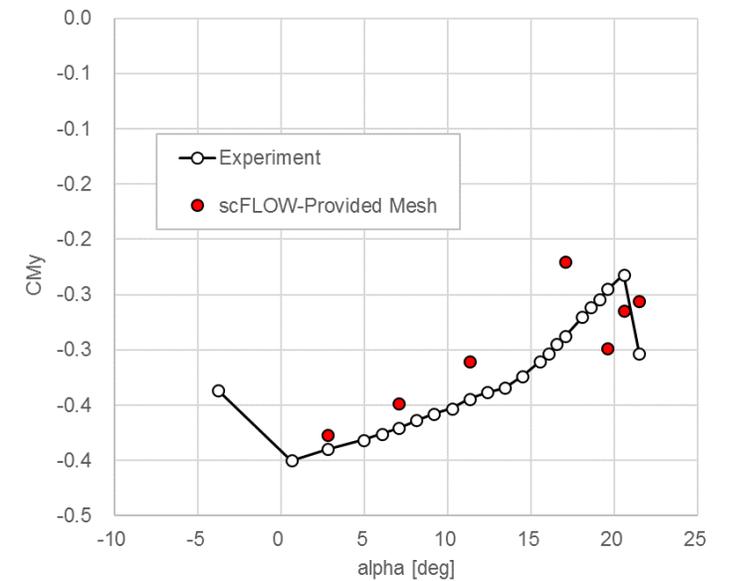
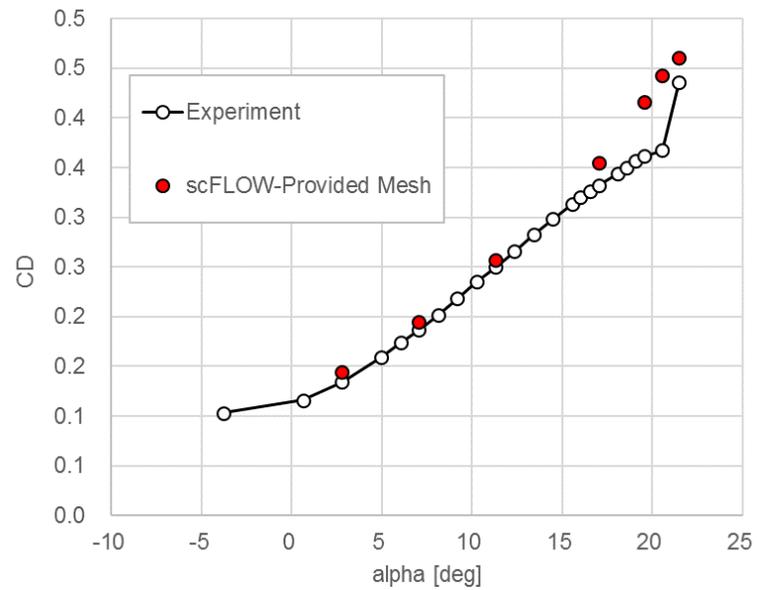
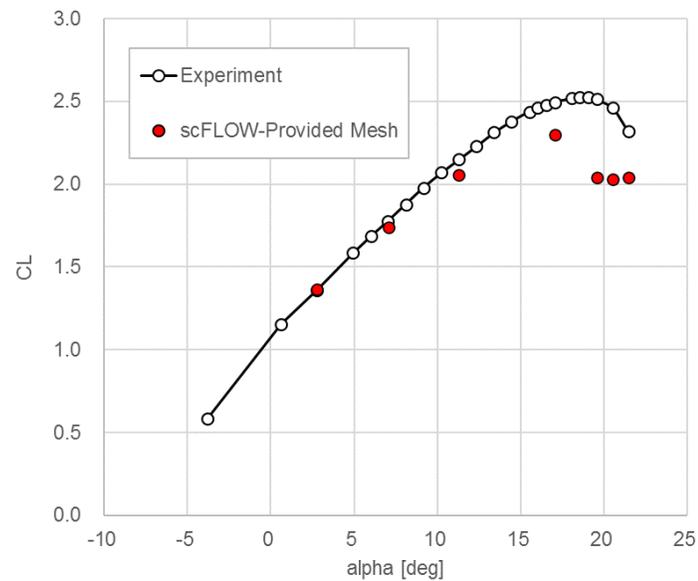
Numerical Conditions

- Mesh HLPW-4_CRM-HL_40-37_Nominal_v1a_Unstr-Hex-Prism-Pyr-Tet_Level_A_PW_v2
 - Total elements 21,868,681
 - Total nodes 12,332,217
 - Hexahedra 9,441,360
 - Tetrahedra 9,428,180
 - Prisms 834,536
 - Pyramids 21,164,605
- Attack of angle [deg] 2.78, 7.05, 11.29, 17.05, 19.57, 20.55, 21.47
- Calculate mode Steady State
- Calculate cycles 20000

Result of mesh adaptation

Aerodynamics coefficients

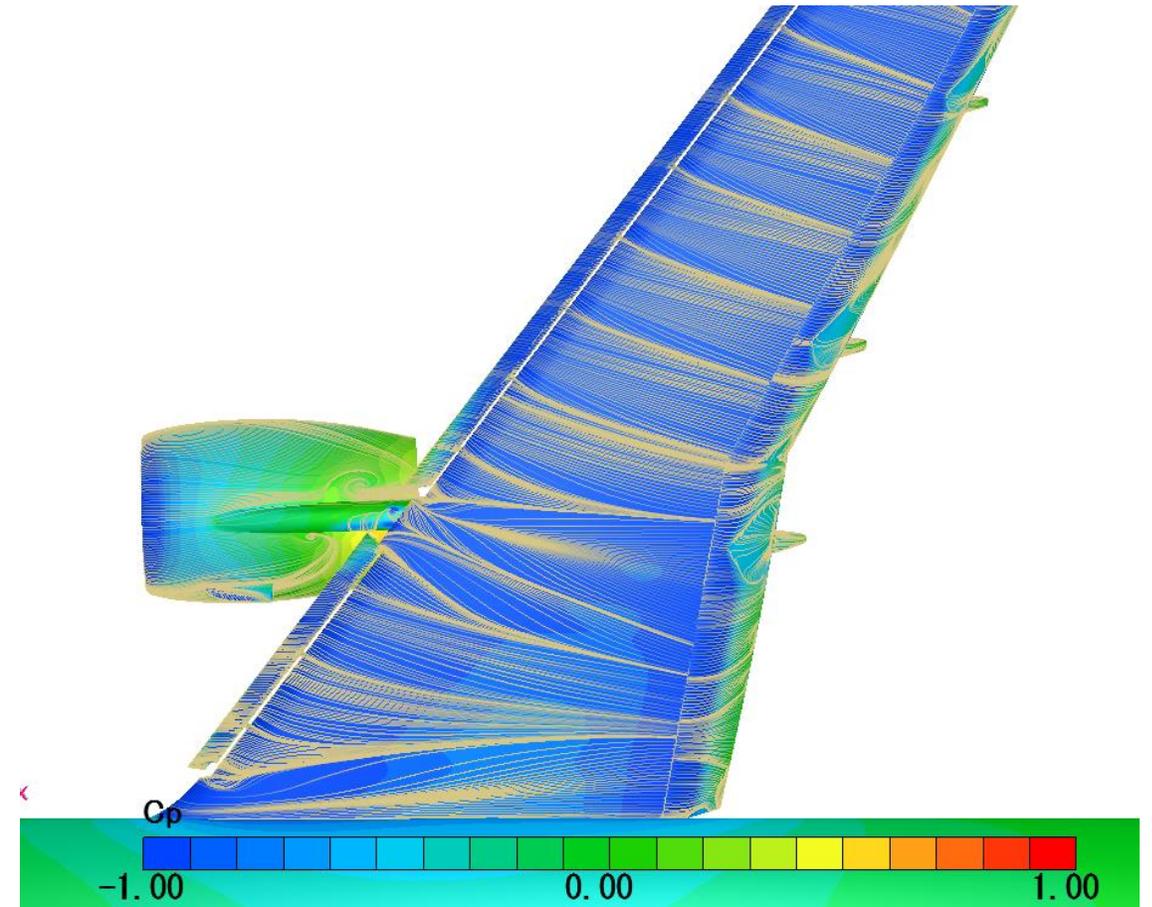
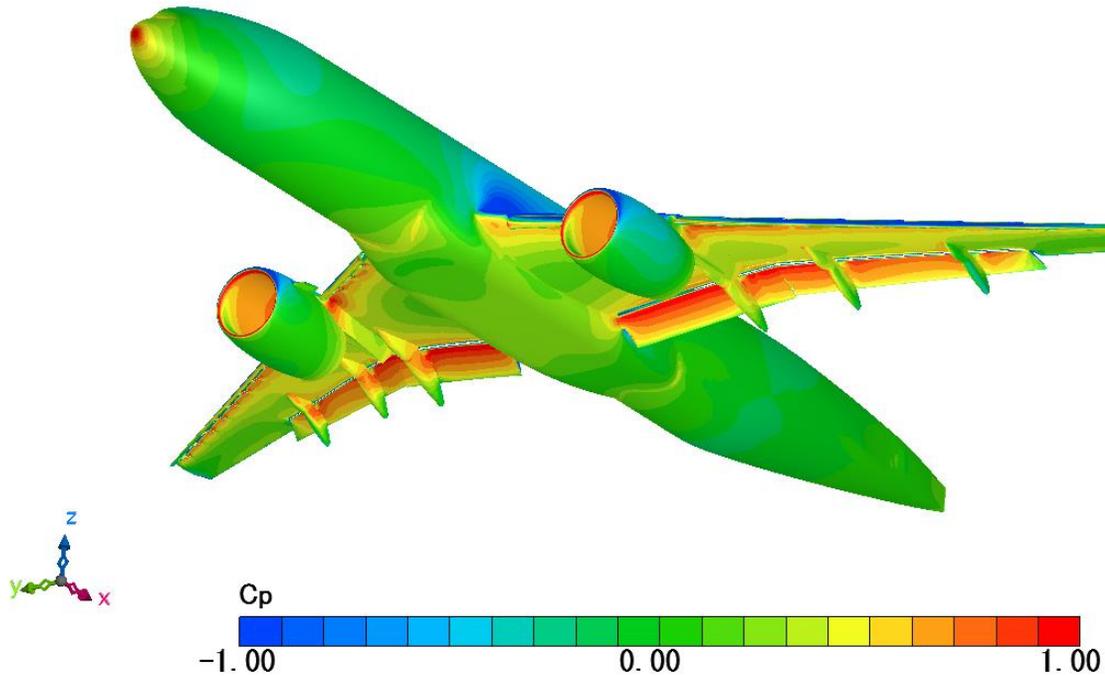
- Lift, drag, and momentum coefficients
- Forces and moments Average for last 1000 cycles



Result of mesh adaptation

Aerodynamics coefficients

The surface pressure coefficient and oil flow ($\alpha = 7.05[\text{deg}]$)



Conclusion

Current progress and future works

- Current progress
 - We calculated the change in performance due to the angle of attack using the provided mesh.
 - In the angle of attack is small, the results close to the experimental results were obtained.
- Future works
 - We create a polyhedral mesh and compare it with the provided mesh and experimental results.